

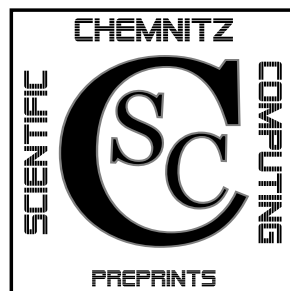


TECHNISCHE UNIVERSITÄT CHEMNITZ

Michael Weise

A framework for efficient hierarchic plate
and shell elements

CSC/17-01



Chemnitz Scientific Computing
Preprints

Impressum:

Chemnitz Scientific Computing Preprints — ISSN 1864-0087

(1995–2005: Preprintreihe des Chemnitzer SFB393)

Herausgeber:

Professuren für
Numerische und Angewandte Mathematik
an der Fakultät für Mathematik
der Technischen Universität Chemnitz

Postanschrift:

TU Chemnitz, Fakultät für Mathematik
09107 Chemnitz

Sitz:

Reichenhainer Str. 41, 09126 Chemnitz

<http://www.tu-chemnitz.de/mathematik/csc/>



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Abstract

The Mindlin–Reissner plate model is widely used for the elastic deformation simulation of moderately thick plates. Shear locking occurs in the case of thin plates, which means slow convergence with respect to the mesh size. The Kirchhoff plate model does not show locking effects, but is valid only for thin plates. One would like to have a method suitable for both thick and thin plates.

Several approaches are known to deal with the shear locking in the Mindlin–Reissner plate model. In addition to the well known MITC elements and other approaches based on a mixed formulation, hierarchic methods have been developed in the recent years. These are based on the Kirchhoff model and add terms to account for shear deformations.

We present some of these methods and develop a new hierarchic plate formulation. This new model can be discretised by a combination of C^0 and C^1 finite elements. Numerical tests show that the new formulation is locking free and numerically efficient.

We also give an extension of the model to a hierarchic Naghdi shell based on a Koiter shell formulation with unknowns in Cartesian coordinates.

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1 Introduction

The Mindlin–Reissner plate model is widely used for the elastic deformation simulation of moderately thick plates. The weak formulation of the model features functions from H^1 and can thus be discretised with C_0 finite elements. This simple approach works if the considered plate is rather thick, but it leads to problems when used with thin plates. One observes very slow convergence with respect to the mesh size. Meshes sufficiently fine for a thick plate may yield results which are several orders of magnitude too small in case of a thin plate. This effect is called shear locking in the engineering literature due to the fact that the plate seems to be stiffer than it is with an insufficiently fine mesh.

On the other hand, the Kirchhoff plate model does not show such locking effects, but is valid only for thin plates. It excludes out-of-plane shear deformations, which are negligible for thin plates but relevant for thick plates. The weak formulation features functions from H^2 and therefore requires C_1 finite elements for a conforming discretisation.

One would like to have a method suitable for both thick and thin plates. Several approaches are known to deal with the shear locking in the Mindlin–Reissner model. In addition to the well known MITC elements and other approaches based on a mixed formulation, hierarchic methods have been developed in the recent years. We focus on hierarchic methods in this article and present two new formulations. In fact we will not really discuss specific elements, but different formulations of the plate theory which may then be discretised with suitable elements.

We will present and discuss numerical examples achieved with two combinations of C_0 and C_1 elements: linear Lagrange ansatz functions and reduced Hsieh–Clough–Tocher (rHCT) ansatz functions for triangular elements and bilinear Lagrange ansatz functions and Bogner–Fox–Schmit (BFS) ansatz functions for rectangular elements. See for example [7], [1], [17] for details on rHCT elements and [3], [13] for details on BFS elements.

The article is structured as follows. Section 2 deals with plate formulations. A simple benchmark problem is given in 2.1. Subsection 2.2 introduces some basic concepts and formulas for the plate problem. Subsections 2.3 to 2.10 present known and new plate formulations and give a short assessment of their performance with conforming elements based on the benchmark problem from 2.1. A more thorough numerical comparison is presented in subsection 2.11. Section 3 presents the basic concepts of the Naghdi and Koiter shell theories in 3.1 and 3.2 and extends the new plate formulations to shells in 3.3 and 3.4 with numerical examples in 3.5. The article is concluded in section 4.

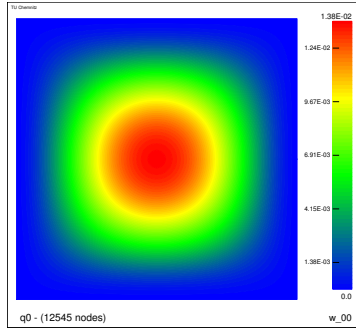


Figure 1: Plate deflection under load

2 Plate theory

2.1 A simple benchmark problem

Before presenting the different plate formulations we define a benchmark problem which will be used for a first assessment of each theory in the following sections. A deeper analysis and comparison of the numerical performance is then given in section 2.11.

We consider a square plate of length 1 with isotropic material, i. e. same material behaviour in all space directions. All edges are hard clamped which means no deflection and no bending angle. This reads $w = \theta = 0$ in the variables defined in the following section. A thick plate with thickness $d = 10^{-1}$ and a thin plate with thickness $d = 10^{-3}$ are subjected to a scaled load of d^3 . See Figure 1 for the qualitative resulting deflection in the thin plate case. Rough reference solutions for the maximum deflection are 1.60×10^{-2} for the thick plate and 1.38×10^{-2} for the thin plate.

The first assessment is done with uniform refinements of one quadratic finite element comprising the whole plate as initial coarse mesh. The full comparison in section 2.11 also features triangular elements. C_0 linear Lagrange ansatz functions and C_1 reduced Hsieh–Clough–Tocher ansatz functions are used for triangular elements and C_0 bilinear Lagrange ansatz functions and C_1 Bogner–Fox–Schmit ansatz functions are used for quadratic elements.

The FE system is solved with the preconditioned conjugate gradient method. Hierarchic preconditioning is used, see [11]. The relative decrease of the residual in the preconditioned norm

$$(r_k^\top w_k)^{1/2} < tol (r_0^\top w_0)^{1/2}$$

with residuals and preconditioned residuals r_k, w_k of iteration k and $tol = 10^{-4}$ serves as stopping criterion. The initial residual r_0 is computed from a zero solution on the initial

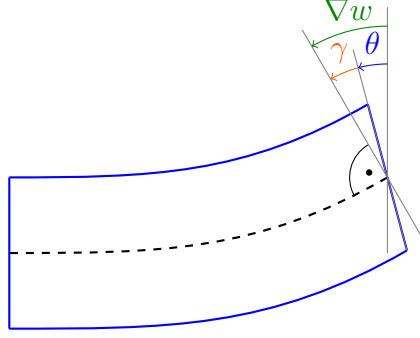


Figure 2: Deformed plate from one side; thickness exaggerated

coarse mesh and from an interpolated solution of the next coarser mesh on all refined meshes.

2.2 Basic assumptions and formulas

Figure 2 depicts a deformed plate with extremely exaggerated thickness viewed from the side. The Mindlin–Reissner hypothesis states that a straight line vertical to the plate midsurfaces remains a straight line after deformation, possibly with a different angle than before. The angle between the original line and the same line after deformation is called the bending angle θ . We abbreviate the spatial derivative $\partial \bullet / \partial x_i$ of any object \bullet with an index \bullet_i throughout this section. The 2D gradient of the vertical deflection w , $\nabla w = [w_{,1}, w_{,2}]^\top = [\partial w / \partial x_1, \partial w / \partial x_2]^\top$, can also be viewed as an angle. This allows for the definition of the shear angle γ by $\gamma = \nabla w - \theta$ or, equivalently,

$$\nabla w = \theta + \gamma. \quad (1)$$

The 3D deformation of a point in the plate domain $(\eta^1, \eta^2, \tau) \in \omega \times [-d/2, d/2]$, $\omega \subset \mathbb{R}^2$ under the Mindlin–Reissner hypothesis may then be described by

$$u^{3D}(x_1, x_2, x_3) = \begin{bmatrix} -x_3 \theta_1(x_1, x_2) \\ -x_3 \theta_2(x_1, x_2) \\ w(x_1, x_2) \end{bmatrix}. \quad (2)$$

We collect the unknowns for any of the following plate formulations in a vector called u . Plugging the deformation ansatz (2) into the 3D elasticity bilinear form and integrating over the thickness direction variable x_3 results in the problem

$$\text{find } u \in \mathbb{V} \text{ with } a(u, \tilde{u}) = l(\tilde{u}) \quad \forall \tilde{u} \in \mathbb{V}_{\text{test}} \quad (3)$$

with the bilinear form

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon(\theta) : \mathbf{c}^b : \varepsilon(\tilde{\theta}) \, d\omega + \int_{\omega} \gamma \cdot \mathbf{c}^s \cdot \tilde{\gamma} \, d\omega, \quad (4)$$

#El	bilin., thick plate		bilin., thin plate		biquadr., thick plate		biquadr., thin plate	
	w_{\max}	#it	w_{\max}	#it	w_{\max}	#it	w_{\max}	#it
4	2.44E-3	2	2.44E-7	2	8.82E-3	4	1.39E-6	4
16	6.67E-3	9	9.26E-7	9	1.49E-2	24	4.95E-6	42
64	1.17E-2	24	3.43E-6	38	1.59E-2	42	4.88E-3	441
256	1.46E-2	40	1.34E-5	114	1.60E-2	42	1.08E-2	1444
1024	1.56E-2	48	5.31E-5	286	1.60E-2	35	1.23E-2	227
4096	1.59E-2	49	2.10E-4	662	1.60E-2	33	1.29E-2	891
16384	1.60E-2	47	8.01E-4	1489	1.60E-2	32	1.32E-2	1426

Table 1: Results for the standard Mindlin–Reissner plate formulation with C^0 elements

the linear form

$$l(\tilde{u}) = \int_{\omega} p \tilde{w} \, d\omega, \quad (5)$$

and appropriate FE ansatz and test spaces \mathbb{V} and \mathbb{V}_{test} . The linearised 2D strain tensor $\varepsilon(\bullet) = \frac{1}{2}(\nabla \bullet + (\nabla \bullet)^{\top})$ is given as the symmetrised 2D gradient. The fourth order material bending tensor \mathfrak{C}^b and the second order shear tensor \mathfrak{C}^s are proportional to d^3 and d , respectively, due to the thickness integration.

The given formulation of the problem and the bilinear and linear forms features the three variables w, θ, γ . The relation $\nabla w = \theta + \gamma$ from (1) allows the elimination of either θ or γ from the problem. This leads to differing formulations with different numerical behaviour which are explored in the following sections.

2.3 The standard Mindlin–Reissner plate formulation (MRs)

The standard formulation of the Mindlin–Reissner plate problem follows from eliminating $\gamma = \nabla w - \theta$ from the three variable formulation (2)–(5). This leads to the bilinear form

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon(\theta) : \mathfrak{C}^b : \varepsilon(\tilde{\theta}) \, d\omega + \int_{\omega} (\nabla w - \theta) \cdot \mathfrak{C}^s \cdot (\nabla \tilde{w} - \tilde{\theta}) \, d\omega$$

with the unknowns $u = [w, \theta_1, \theta_2]^{\top}$. All unknowns are featured with derivatives up to first order and are thus assumed to be in $H^1(\omega)$. This allows for a simple discretisation with C^0 finite elements.

Results for the example problem from section 2.1 with bilinear and biquadratic rectangular elements are shown in Table 1. One observes reasonable h -convergence for the thick plate case but very slow convergence for the thin plate case. This effect is called thickness locking or shear locking. Standard finite elements are not capable of resolving the first (bending) term and the second (shear) term of the bilinear form appropriately for small thickness d . As d tends to zero, the shear term dominates due to the scaling of the bending term with d^3 and the shear term with d . The shear term is imbalanced in the sense that it consists of a function value and derivative value.

#El	thick plate		thin plate	
	w_{\max}	#it	w_{\max}	#it
4	1.45E-2	2	1.45E-2	2
16	1.38E-2	8	1.38E-2	8
64	1.38E-2	23	1.38E-2	22
256	1.38E-2	35	1.38E-2	35
1024	1.38E-2	29	1.38E-2	29
4096	1.38E-2	29	1.38E-2	28
16384	1.38E-2	31	1.38E-2	83

Table 2: Results for the Kirchhoff plate problem with C^1 BFS elements

2.4 The Kirchhoff plate

The plate model of Kirchhoff can be viewed as a special case of the Mindlin–Reissner plate without the allowance for a shear angle. The condition $\gamma = 0$ gives $\theta = \nabla w$ from (1). The geometric interpretation of this ansatz is that orthogonal line elements prior to deformation are still orthogonal to the new plate midsurface after deformation. The problem formulation changes to

$$u^{3D} = \begin{bmatrix} -x_3 w_{,1} \\ -x_3 w_{,2} \\ w \end{bmatrix}, \quad a(u, \tilde{u}) = \int_{\omega} \varepsilon(\nabla w) : \mathfrak{C}^b : \varepsilon(\nabla \tilde{w}) \, d\omega, \quad l(\tilde{u}) = \int_{\omega} p \tilde{w} \, d\omega$$

with the single unknown $u = w$. The bilinear form features second derivatives which calls for $w \in H^2(\omega)$. Thus, C^1 finite elements are needed for a conforming discretisation.

Results for the example problem from section 2.1 with BFS elements are shown in Table 2. Even very coarse meshes yield a quite accurate solution for the thin plate. No shear locking is observed because no shear term is present. The thick plate example converges equally fast, but to the same value of 1.38×10^{-2} while the reference solution for the thick plate is about 1.60×10^{-2} . This is due to the fact that the Kirchhoff model neglects the shear term and is thus only suitable for thin plates.

2.5 The Mindlin–Reissner plate in hierarchic formulation (MRh)

The desire for a suitable plate formulation for thick and thin plates has led to numerous reformulations of the Mindlin–Reissner plate problem with the goal of the exclusion of locking. A hierarchic approach can be found in [8] based on earlier works, see the references therein. The cited work uses isogeometric analysis (finite elements with non-uniform rational B-splines used for geometry definition and as basis functions) but the formulation given there can also be discretised with standard finite elements.

#El	thick plate		thin plate	
	w_{\max}	#it	w_{\max}	#it
4	1.45E-2	2	1.45E-2	2
16	1.53E-2	10	1.38E-2	9
64	1.56E-2	32	1.38E-2	24
256	1.58E-2	73	1.38E-2	45
1024	1.59E-2	130	1.38E-2	68
4096	1.60E-2	214	1.38E-2	92
16384	1.60E-2	423	1.38E-2	108

Table 3: Results for the hierarchic Mindlin–Reissner formulation with C^1 BFS elements for w and C^0 bilinear elements for γ

The idea is to eliminate $\theta = \nabla w - \gamma$ from the three variable formulation (2)–(5) instead of γ . This leads to

$$u^{3D} = \begin{bmatrix} -x_3(w_{,1} - \gamma_1) \\ -x_3(w_{,2} - \gamma_2) \\ w \end{bmatrix},$$

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon(\nabla w - \gamma) : \mathbf{e}^b : \varepsilon(\nabla \tilde{w} - \tilde{\gamma}) \, d\omega + \int_{\omega} \gamma \cdot \mathbf{c}^s \cdot \tilde{\gamma} \, d\omega,$$

$$l(\tilde{u}) = \int_{\omega} p \tilde{w} \, d\omega$$

with the variables $u = [w, \gamma_1, \gamma_2]^T$. First derivatives of γ and second derivatives of w call for $\gamma \in H^1(\omega)$ and $w \in H^2(\omega)$. Thus, a discretisation with C^0 elements for γ and C^1 elements for w is needed. The name “hierarchic” comes from the fact that the bilinear form includes the Kirchhoff bending term and can thus be viewed as a hierarchic extension of the Kirchhoff model.

Results for the example problem from section 2.1 are shown in Table 3. One observes good h -convergence to the right solutions for the thick and the thin plate. The hierarchic reformulation has rendered the problem locking free and the method performs well for thick and thin plates. There is, however, one major drawback: The iteration numbers needed to reduce the initial residual below the defined threshold are significantly higher for the thick plate than for the thin plate and also much higher than with the standard Mindlin–Reissner formulation. A fair and thorough comparison between the pure C^0 elements used there and the partially C^1 combined elements here also needs to involve some error and computation time measurement and is carried out in section 2.11. But the reduced comparison given here still serves to show that it might be useful to search for further formulations even though the hierarchic formulation is locking free.

#El	thick plate		thin plate	
	w_{\max}	#it	w_{\max}	#it
4	1.63E-2	3	1.45E-2	3
16	1.56E-2	34	1.38E-2	14
64	1.58E-2	265	1.38E-2	63
256	1.59E-2	910	1.38E-2	204
1024	1.60E-2	2367	1.38E-2	559
4096	1.60E-2	5000	1.38E-2	1241
16384	1.60E-2	5000	1.38E-2	2837

Table 4: Results for the rotation free Mindlin–Reissner formulation with C^1 BFS elements for w , w_{sb} and w_{bs}

2.6 The Mindlin–Reissner plate in a rotation free formulation by Oesterle, Ramm and Bischoff (ORB)

The locking phenomenon is caused by the imbalance of function values and derivative values in the standard formulation. The authors of [16] try to overcome this issue with a so-called rotation free formulation. All variables of this formulation represent displacements and not rotation angles. Two additional plate deflections whose derivatives represent the shear angles need to be introduced for an equivalent formulation to the Mindlin–Reissner plate. The ansatz

$$w = w_{\text{b}} + w_{\text{sb}} + w_{\text{bs}}, \gamma = (w_{\text{sb},1}, w_{\text{bs},2})^{\text{T}}, \theta = \nabla w - \gamma = \nabla w_{\text{b}} + (w_{\text{bs},1}, w_{\text{sb},2})^{\text{T}}$$

yields the formulation

$$u^{3\text{D}} = \begin{bmatrix} u_1 - x_3(w_{\text{b},1} + w_{\text{bs},1}) \\ u_2 - x_3(w_{\text{b},2} + w_{\text{sb},2}) \\ w \end{bmatrix},$$

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon \begin{pmatrix} w_{\text{b},1} + w_{\text{bs},1} \\ w_{\text{b},2} + w_{\text{sb},2} \end{pmatrix} : \mathbf{c}^{\text{b}} : \varepsilon \begin{pmatrix} \tilde{w}_{\text{b},1} + \tilde{w}_{\text{bs},1} \\ \tilde{w}_{\text{b},2} + \tilde{w}_{\text{sb},2} \end{pmatrix} d\omega + \int_{\omega} \begin{pmatrix} w_{\text{sb},1} \\ w_{\text{bs},2} \end{pmatrix} \cdot \mathbf{c}^{\text{s}} \cdot \begin{pmatrix} \tilde{w}_{\text{sb},1} \\ \tilde{w}_{\text{bs},2} \end{pmatrix} d\omega,$$

$$l(\tilde{u}) = \int_{\omega} p(\tilde{w}_{\text{b}} + \tilde{w}_{\text{sb}} + \tilde{w}_{\text{bs}}) d\omega.$$

All three variables w_{b} , w_{sb} , w_{bs} are present with second derivatives and need to be discretised with C^1 elements. Unlike the source [16] which uses isogeometric analysis we again employ standard C^1 finite elements.

Results for the example problem from section 2.1 are shown in Table 4. The conjugate gradient method was stopped if the stopping criterion was not met after 5000 iterations. One observes good h -convergence to the right solutions for the thick and the thin plate. The rotation free reformulation has rendered the problem locking free and the method performs well for thick and thin plates. The iteration numbers, however, are extremely high in comparison with the other methods. With shear locking gone but very low

#El	thick plate		thin plate	
	w_{\max}	#it	w_{\max}	#it
4	1.63E-2	3	1.45E-2	3
16	1.55E-2	24	1.38E-2	119
64	1.56E-2	49	1.38E-2	1320
256	1.57E-2	96	1.38E-2	3132
1024	1.57E-2	103	1.38E-2	2666
4096	1.57E-2	139	1.38E-2	834
16384	1.57E-2	142	1.38E-2	483

Table 5: Results for the Endo–Kimura plate formulation with C^1 BFS elements for w and w_b

numerical efficiency this method does not look promising for practical use. Maybe there exist better solvers for this problem formulation, but at least our preconditioned solver is inefficient in this case.

2.7 The plate formulation of Endo and Kimura (EKs)

The authors of [9] argue that the bending and shear deformations of the Mindlin–Reissner plate formulation can not be determined uniquely and therefore propose a different formulation. Like in the standard formulation they eliminate $\gamma = \nabla w - \theta$ from the three variable formulation (2)–(5) and make the additional assumption $\theta = \nabla w_b$ for some w_b . In consequence it holds $\gamma = \nabla w_s$ for $w_s = w - w_b$. Just like the formulation of the previous section the absence of rotation angle variables leads to a rotation free and thus also locking free formulation. One gets

$$u^{3D} = \begin{bmatrix} u_1 - x_3 w_{b,1} \\ u_2 - x_3 w_{b,2} \\ w \end{bmatrix},$$

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon(\nabla w_b) : \mathfrak{C}^b : \varepsilon(\nabla \tilde{w}_b) \, d\omega + \int_{\omega} (\nabla w - \nabla w_b) \cdot \mathfrak{C}^s \cdot (\nabla \tilde{w} - \nabla \tilde{w}_b) \, d\omega,$$

$$l(\tilde{u}) = \int_{\omega} p \tilde{w} \, d\omega.$$

Both variables w and w_b are present with second derivatives and need to be discretised with C^1 elements. One expects a slightly different solution than with the standard Mindlin–Reissner plate due to the additional condition that θ needs to be a gradient.

Numerical results are shown in Table 5. One observes good h -convergence for both the thick and the thin plate. The solution of the thick plate is different from the Mindlin–Reissner plate. The additional gradient condition leads to a slightly stiffer behaviour. It is unclear which solution is the “better” one compared to the behaviour of real plates. Iteration numbers are good for the thin plate but very high for the thick plate.

#El	thick plate		thin plate	
	w_{\max}	#it	w_{\max}	#it
4	1.63E-2	3	1.45E-2	3
16	1.55E-2	17	1.38E-2	9
64	1.56E-2	103	1.38E-2	28
256	1.57E-2	374	1.38E-2	48
1024	1.57E-2	706	1.38E-2	59
4096	1.57E-2	1383	1.38E-2	100
16384	1.57E-2	2829	1.38E-2	139

Table 6: Results for the hierarchic Endo–Kimura plate formulation with C^1 BFS elements for w and w_s

2.8 The Endo–Kimura plate in hierarchic formulation (EKh)

Endo and Kimura also suggested that their ansatz $\theta = \nabla w_b, \gamma = w_s$ can be combined with the hierarchic ansatz of eliminating θ instead of γ from the three variable formulation (2)–(5). This yields

$$u^{3D} = \begin{bmatrix} u_1 - x_3(w_{,1} - w_{s,1}) \\ u_2 - x_3(w_{,2} - w_{s,2}) \\ w \end{bmatrix},$$

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon(\nabla w - \nabla w_s) : \mathbf{e}^b : \varepsilon(\nabla \tilde{w} - \nabla \tilde{w}_s) d\omega + \int_{\omega} (\nabla w_s) \cdot \mathbf{e}^s \cdot (\nabla \tilde{w}_s) d\omega,$$

$$l(\tilde{u}) = \int_{\omega} p \tilde{w} d\omega.$$

The numerical results in Table 6 show good h -convergence to the same solution like the standard Endo–Kimura formulation. Iteration numbers are good for the thick plate but very high for the thin plate. This is exactly the other way around than with standard Endo–Kimura in the previous section. It appears that the difficulty of the problem has shifted from the thick to the thin plate, or from the shear to the bending term.

2.9 First new formulation: Endo–Kimura plate decoupled (EKd)

Another kind of elimination which was not present in [9] is possible. With the ansatz of Endo and Kimura one can also eliminate $\nabla w = \nabla w_b + \nabla w_s$ and $w = w_b + w_s$ from the

#El	thick plate		thin plate	
	w_{\max}	#it	w_{\max}	#it
4	1.69E-2	2	1.45E-2	2
16	1.58E-2	10	1.38E-2	9
64	1.58E-2	22	1.38E-2	22
256	1.57E-2	26	1.38E-2	35
1024	1.57E-2	24	1.38E-2	29
4096	1.57E-2	21	1.38E-2	29
16384	1.57E-2	20	1.38E-2	33

Table 7: Results for the decoupled Endo–Kimura plate formulation with C^1 BFS elements for w_b and C^0 bilinear elements for w_s

equations. Only the variables w_b and w_s remain and one arrives at

$$u^{3D} = \begin{bmatrix} u_1 - x_3 w_{b,1} \\ u_2 - x_3 w_{b,2} \\ w \end{bmatrix},$$

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon(\nabla w_b) : \mathbf{c}^b : \varepsilon(\nabla \tilde{w}_b) d\omega + \int_{\omega} (\nabla w_s) \cdot \mathbf{c}^s \cdot (\nabla \tilde{w}_s) d\omega,$$

$$l(\tilde{u}) = \int_{\omega} p(\tilde{w}_b + \tilde{w}_s) d\omega.$$

This formulation has completely decoupled bending and shear terms. The bending deflection w_b is the same as in the Kirchhoff theory and an additional shear deflection w_s may be calculated separately and added for a total deflection $w = w_b + w_s$ afterwards. We solve, however, the complete system. An analogous decoupled formulation is not directly obtainable from the standard Mindlin–Reissner formulation. A conforming discretisation can be achieved with C^1 elements for w_b and C^0 elements for w_s .

The numerical results in Table 7 show good h -convergence to the same solution like the standard Endo–Kimura formulation. Iteration numbers are very good for both the thick and the thin plate.

The decoupling has eliminated the convergence problems of the other two Endo–Kimura formulations. We have obtained a formulation which is suitable for both thick and thin plates and is numerically efficient in both cases. The only drawback is that we do not get the solution of the Mindlin–Reissner plate but of a slightly stiffer problem.

2.10 Second new formulation: hierarchic Mindlin–Reissner based on Endo–Kimura (MREK)

We propose to extend the formulation of the previous section with an additional term to relax the gradient condition again. In this way we arrive at a Mindlin–Reissner

#El	thick plate		thin plate	
	w_{\max}	#it	w_{\max}	#it
4	1.69E-2	2	1.45E-2	2
16	1.60E-2	9	1.38E-2	10
64	1.60E-2	26	1.38E-2	25
256	1.60E-2	46	1.38E-2	50
1024	1.60E-2	60	1.38E-2	68
4096	1.60E-2	100	1.38E-2	100
16384	1.60E-2	130	1.38E-2	132

Table 8: Results for the Endo–Kimura based Mindlin–Reissner plate formulation with C^1 BFS elements for w_b and C^0 bilinear elements for w_s and ρ

formulation which can be viewed as hierarchically based on the decoupled Endo–Kimura formulation. The ansatz

$$w = w_b + w_s, \quad \nabla w = \theta + \gamma, \quad \theta = \nabla w_b - \rho, \quad \gamma = \nabla w_s + \rho$$

and elimination of w yields

$$u^{3D} = \begin{bmatrix} u_1 - x_3(w_{b,1} - \rho_1) \\ u_2 - x_3(w_{b,2} - \rho_2) \\ w \end{bmatrix},$$

$$a(u, \tilde{u}) = \int_{\omega} \varepsilon(\nabla w_b - \rho) : \mathfrak{C}^b : \varepsilon(\nabla \tilde{w}_b - \tilde{\rho}) \, d\omega + \int_{\omega} (\nabla w_s + \rho) \cdot \mathfrak{C}^s \cdot (\nabla \tilde{w}_s + \tilde{\rho}) \, d\omega,$$

$$l(\tilde{u}) = \int_{\omega} p(\tilde{w}_b + \tilde{w}_s) \, d\omega.$$

A conforming discretisation can be achieved with C^1 elements for w_b and C^0 elements for ρ_1 , ρ_2 and w_s .

The numerical results in Table 8 show good h -convergence to the same solution like for the Mindlin–Reissner model. Iteration numbers are not as good as for the decoupled Endo–Kimura formulation but still good. They are comparable for the thick and thin plate and grow slowly with the number of elements. In comparison with the hierarchic Mindlin–Reissner formulation from 2.5, the iteration numbers for the thin plate are slightly bigger and those of the thick plate are somewhat higher at lower element numbers but are growing slower and thus they are lower than those of the hierarchic Mindlin–Reissner formulation at higher element numbers.

2.11 Comparison of numerical results

The numerical example from section 2.1 is explored in more detail in this section. We consider the error estimator of [4] for MITC elements and neglect the terms which are

formulation	C^0 unkn.	C^1 unkn.	dofs/node (rectangle)	dofs/node (triangle)
MRs	3	0	3	3
MRh	2	1	6	5
ORB	0	3	12	9
EKs	0	2	8	6
EKh	0	2	8	6
EKd	1	1	5	4
MREK	3	1	7	6

Table 9: Continuous and discretised unknowns of the plate formulations

zero for our standard elements. The error contribution of an element T then reads

$$\begin{aligned} \eta_T^2 = & h_T^2 \left(\|\operatorname{div}(\mathbf{c}^b : \varepsilon(\theta_h)) + \mathbf{c}^s \cdot \gamma_h\|_{L^2(T)} + (d^2 + h_T^2) \|p + \operatorname{div}(\mathbf{c}^s \cdot \gamma_h)\|_{L^2(T)} \right) \\ & + \frac{1}{2} \sum_{E \in \mathcal{E}(T)} h_E \left(\|[\mathbf{c}^b : \varepsilon(\theta_h)]_E \cdot \mathbf{n}_E\|_{L^2(E)} + (d^2 + h_E^2) \|[\mathbf{c}^s \cdot \gamma_h]_E \cdot \mathbf{n}_E\|_{L^2(E)} \right) \end{aligned}$$

for a finite element solution θ_h and γ_h (adapted to the actual used theory for each of the Mindlin–Reissner and Endo–Kimura plate formulations). The formula involves the element diameter h_T , the set of edges $\mathcal{E}(T)$ of the element T , the edge length h_E , the jump $[\bullet]_E$ of a quantity over the edge E and a fixed unit normal \mathbf{n}_E of edge E .

Computations were done with our TU Chemnitz adaptive FEM software SPC written in Fortran, see [12]. The software features adaptive FEM but uniform refinement was employed for this tests for good comparability. The plate module of SPC features rectangular and triangular elements. The coarse mesh for our example with rectangular elements is one unit square; the coarse mesh for triangular elements is the unit square divided by one of its diagonals. Square element use Bogner–Fox–Schmit (BFS) quadrangles for C^1 parts and Lagrange elements for C^0 parts of the different formulations as detailed in the sections above. Triangular elements use reduced Hsieh–Clough–Tocher (rHCT) elements for C^1 parts and Lagrange elements for C^0 parts of the different formulations as detailed in the sections above. BFS elements feature 4 degrees of freedom (dofs) per node and unknown in the ansatz, rHCT elements 3 dofs per node and unknown. This leads to unknowns per node as given in Table 9.

Figure 3 collects the development of the error estimator for the thick and thin plates in case of square elements. The first row shows its reduction over the number of elements. All formulations show h -convergence with best results for MREK and EKd for the thick plate and MREK for the thin plate.

The second row shows the error reduction over the total number of unknowns. The unknowns are approximately linked to the element numbers by a fixed factor because of the uniform refinement. Thus, the graphs do look quite similar for both the thick and the thin plate compared to those of the first row.

The third row shows the error reduction over the total time summed over all refinements. These two are the most important graphs because only they factor in the iteration numbers via the computation time. EKd is best closely followed by MREK and they are clearly more efficient than the other formulations for the thick plate. For the thin plate this changes to MREK as the most efficient formulation followed by EKd and then the other formulations.

The results for triangular elements are collected in Figure 4. The performance differences are less pronounced in this case than in the square elements case. EKd is the most efficient formulation for both the thick and the thin plate. MREK, EK_h and MR_h are also performing quite well in both cases. EK_s is very good for thick plates but rather inefficient for the thin plate.

In summary, MREK is the most efficient Mindlin–Reissner formulation and EKd the most efficient Endo–Kimura formulation for thick and thin as well as square and triangular elements. We conclude that both our new formulations are the formulations of choice for flexibly usable thick and thin Mindlin–Reissner or Endo–Kimura plate elements. The MR_h formulation comes close in performance to MREK in case of triangular elements but is clearly outperformed by MREK in case of square elements.

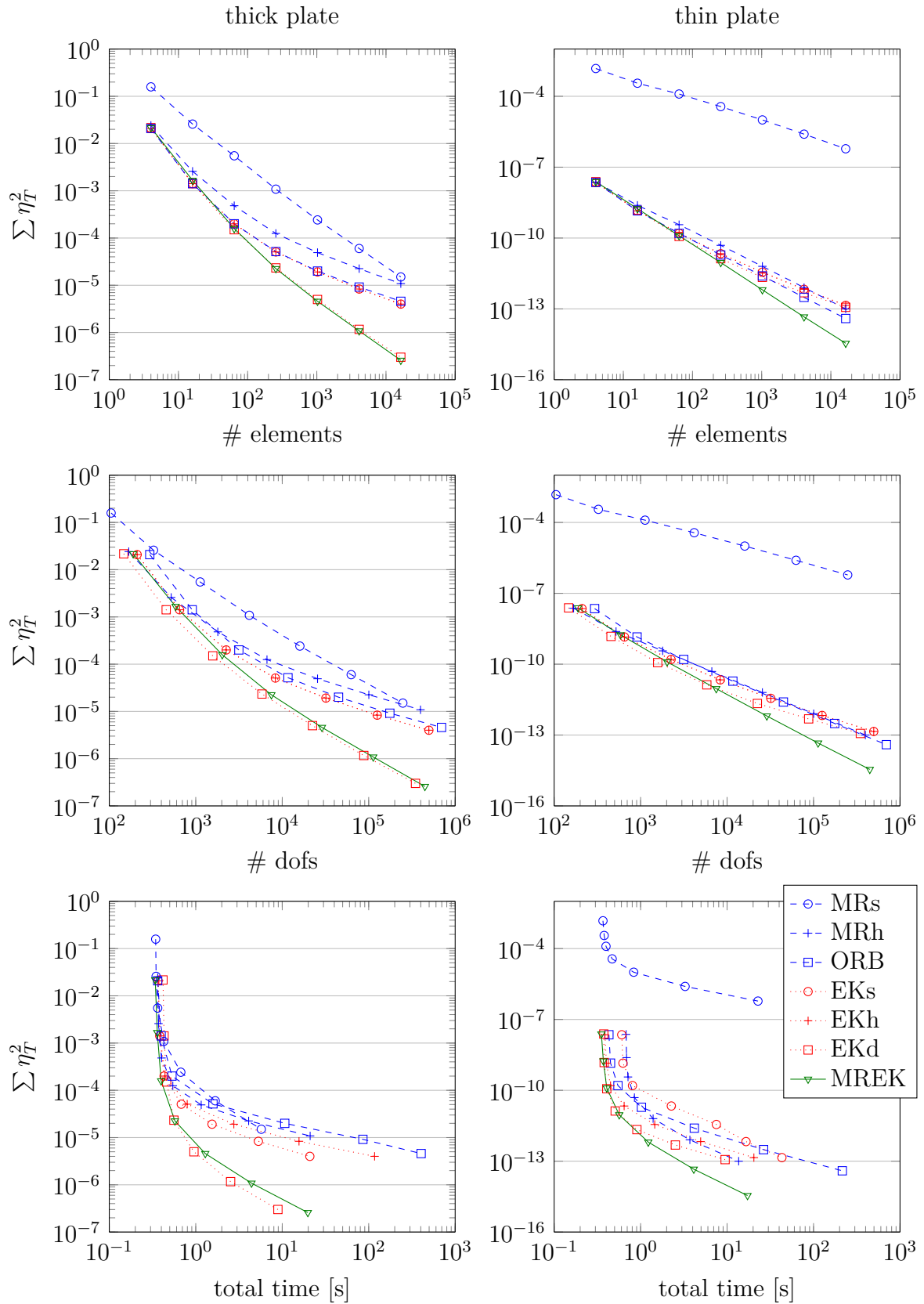


Figure 3: Development of squared relative error estimator for square elements

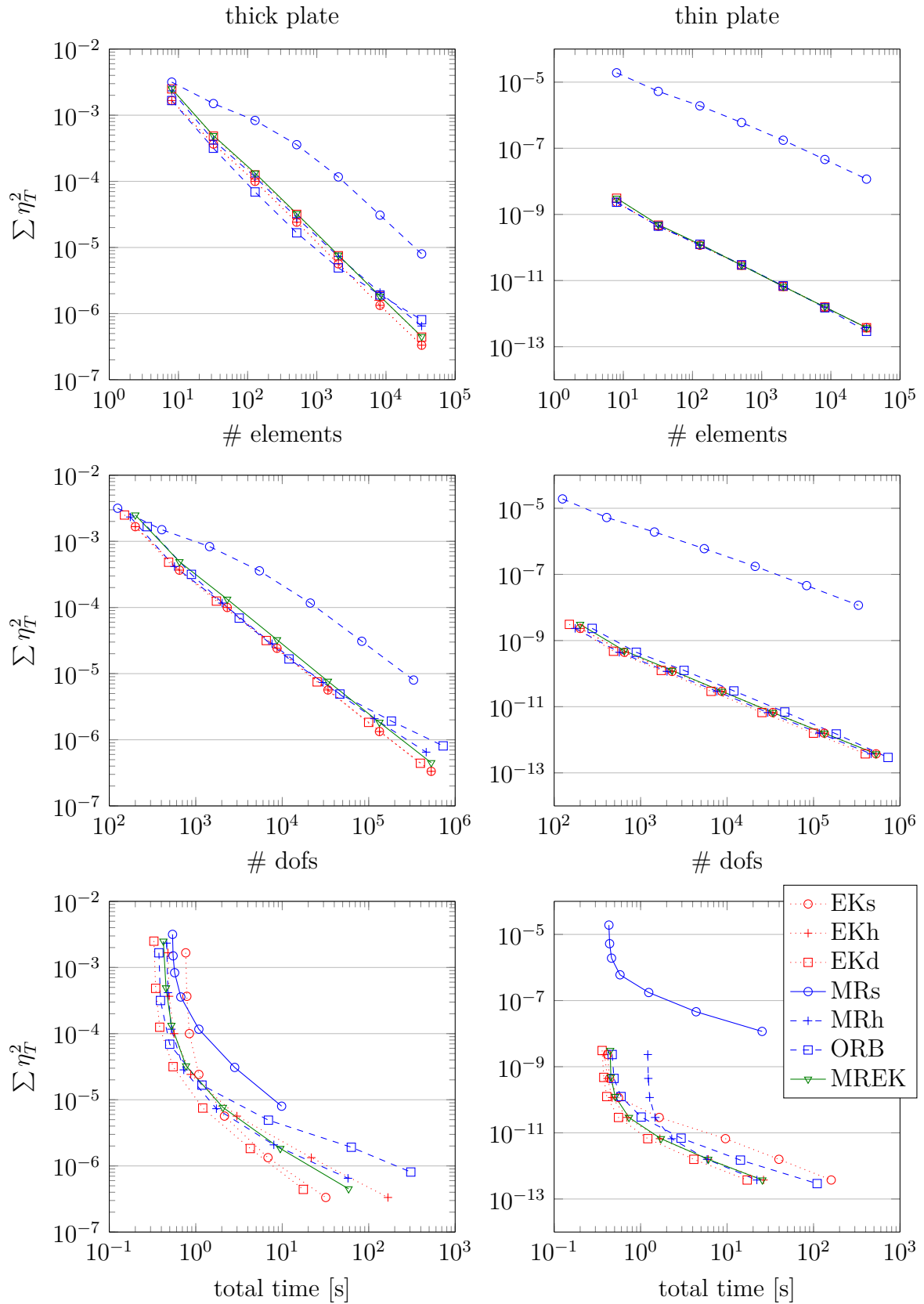


Figure 4: Development of squared relative error estimator for triangle elements

3 Shell theory

3.1 The Naghdi shell

We consider a shell, a thin-walled, possibly curved structure in 3D. The shell domain is defined by the midsurface \mathbf{y} and the thickness d (significantly smaller than the midsurface dimensions) via

$$\{\mathbf{x}(\eta^1, \eta^2, \tau) = \mathbf{y}(\eta^1, \eta^2) + \tau \mathbf{a}_3(\eta^1, \eta^2) : (\eta^1, \eta^2) \in \omega \subset \mathbb{R}^2, \tau \in [-\frac{d}{2}, \frac{d}{2}]\}$$

with the tangential vectors $\mathbf{a}_1, \mathbf{a}_2$ and the unit normal vector \mathbf{a}_3 of the midsurface given by

$$\mathbf{a}_i = \mathbf{y}_{,i} = \frac{\partial \mathbf{y}}{\partial \eta^i}, \quad i = 1, 2, \quad \mathbf{a}_3 = \frac{\mathbf{a}_1 \times \mathbf{a}_2}{\|\mathbf{a}_1 \times \mathbf{a}_2\|}.$$

These vectors are called covariant basis and form a biorthogonal system with the contravariant basis $\mathbf{a}^1, \mathbf{a}^2$ and $\mathbf{a}^3 = \mathbf{a}_3$. We abbreviate the parameter derivative $\partial \bullet / \partial \eta^i$ of any object \bullet with an index $\bullet_{,i}$ throughout this section.

The solution of a shell deformation problem with 3D-FEM is not suitable due to the small thickness (either elements with bad aspect ratios occur or an extreme amount of elements is required). Therefore, special shell models are needed. One popular example is the Naghdi shell model, see for example [5, section 4.2.2] and the references therein. It employs the Mindlin–Reissner hypotheses (normal lines remain straight after deformation, no change of thickness, plane state of stress) with the deformation ansatz

$$\mathbf{u}^{3D} := \mathbf{u}(\eta^1, \eta^2) + \tau \boldsymbol{\theta}(\eta^1, \eta^2) \quad \text{with} \quad \mathbf{u} = u_1 \mathbf{a}^1 + u_2 \mathbf{a}^2 + u_3 \mathbf{a}_3, \quad \boldsymbol{\theta} = \theta_1 \mathbf{a}^1 + \theta_2 \mathbf{a}^2$$

in covariant coordinates with the midsurface translation \mathbf{u} and the rotation $\boldsymbol{\theta}$.

3.1.1 Naghdi shell in covariant coordinates

We collect the five unknown covariant coordinates in a vector

$$u = [u_1, u_2, u_3, \theta_1, \theta_2]^T.$$

The weak formulation of the shell problem reads

$$a(u, \tilde{u}) = f(\tilde{u}) \quad \forall \tilde{u}$$

with test functions \tilde{u} from an appropriate space, the bilinear form $a(\cdot, \cdot)$ accounting for the inner virtual work of the elastic deformation and the linear functional $f(\cdot)$ accounting

for the load of the test functions. The bilinear form can be written as a sum of membrane, bending and shear terms with according strains in the form

$$\begin{aligned}
a(u, \tilde{u}) &= a^m(u, \tilde{u}) + a^b(u, \tilde{u}) + a^s(u, \tilde{u}), \\
a^m(u, \tilde{u}) &= \int_{\omega} \boldsymbol{\varepsilon}^m(u) : \boldsymbol{\mathfrak{C}}^m : \boldsymbol{\varepsilon}^m(\tilde{u}) \, d\mathcal{S}, \\
a^b(u, \tilde{u}) &= \int_{\omega} \boldsymbol{\varepsilon}^b(u) : \boldsymbol{\mathfrak{C}}^b : \boldsymbol{\varepsilon}^b(\tilde{u}) \, d\mathcal{S}, \\
a^s(u, \tilde{u}) &= \int_{\omega} \boldsymbol{\varepsilon}^s(u) : \boldsymbol{\mathfrak{C}}^s : \boldsymbol{\varepsilon}^s(\tilde{u}) \, d\mathcal{S}
\end{aligned} \tag{6}$$

with the surface element $d\mathcal{S} = \|\mathbf{a}_1 \times \mathbf{a}_2\| d\eta^1 d\eta^2$. The membrane, bending and shear strain tensors

$$\boldsymbol{\varepsilon}^m(u) = \Sigma_{ij} \varepsilon_{ij}^m(u) \mathbf{a}^i \mathbf{a}^j, \quad \boldsymbol{\varepsilon}^b(u) = \Sigma_{ij} \varepsilon_{ij}^b(u) \mathbf{a}^i \mathbf{a}^j, \quad \boldsymbol{\varepsilon}^s(u) = \Sigma_i \varepsilon_i^s(u) (\mathbf{a}^i \mathbf{a}_3 + \mathbf{a}_3 \mathbf{a}^i)$$

have the covariant coordinates

$$\begin{aligned}
\varepsilon_{ij}^m(u) &= \frac{1}{2}(u_{i|j} + u_{j|i}) - b_{ij} u_3 \quad (\text{not dependent on } \theta_i), \\
\varepsilon_{ij}^b(u) &= \frac{1}{2}(\theta_{i|j} + \theta_{j|i} - \Sigma_k b_i^k u_{k|j} - \Sigma_k b_j^k u_{k|i}) + c_{ij} u_3, \\
\varepsilon_i^s(u) &= \frac{1}{2}(\theta_i + u_{3,i} + \Sigma_j b_i^j u_j).
\end{aligned} \tag{7}$$

The notation above uses the first fundamental form $a_{ij} = \mathbf{a}_i \cdot \mathbf{a}_j$ or $a^{ij} = \mathbf{a}^i \cdot \mathbf{a}^j$, the second fundamental form $b_{ij} = \mathbf{a}_{i,j} \cdot \mathbf{a}_3 = -\mathbf{a}_i \cdot \mathbf{a}_{3,j}$ or $b_i^j = \Sigma_k b_{ik} a^{kj}$, the third fundamental form $c_{ij} = \Sigma_l b_i^l b_{lj} = \Sigma_{kl} b_{ik} a^{kl} b_{lj}$ and the covariant derivative $u_{i|j} = u_{i,j} - \Sigma_k \Gamma_{ij}^k u_k$ with the Christoffel symbols $\Gamma_{ij}^k = \mathbf{a}_{i,j} \cdot \mathbf{a}^k = \Sigma_l a^{kl} (a_{il,j} + a_{jl,i} - a_{ij,l})/2$. Indices take the values 1 and 2 whenever a (multi-)sum Σ without an index range appears. The membrane and bending material tensors $\boldsymbol{\mathfrak{C}}^m$ and $\boldsymbol{\mathfrak{C}}^b$ are obtained by integrating the material tensor $\boldsymbol{\mathfrak{C}}$ and $\tau^2 \boldsymbol{\mathfrak{C}}$, respectively, over the thickness of the shell. If the material is constant over the thickness this leads to $\boldsymbol{\mathfrak{C}}^m = d\boldsymbol{\mathfrak{C}}$, $\boldsymbol{\mathfrak{C}}^b = \frac{d^3}{12} \boldsymbol{\mathfrak{C}}$. For a complete derivation of the above formulas we again refer to [5].

The action of the second order tensor $\boldsymbol{\varepsilon}^s(u) = \Sigma_i \varepsilon_i^s(u) (\mathbf{a}^i \mathbf{a}_3 + \mathbf{a}_3 \mathbf{a}^i)$ with the fourth order material tensor $\boldsymbol{\mathfrak{C}}^m$ may be expressed by a replacement first order tensor $2\mathbf{e}^s(u) = \Sigma_i \varepsilon_i^s(u) \mathbf{a}^i$ with an appropriately chosen reduced second order shear tensor $\boldsymbol{\mathfrak{C}}^s$ which reads

$$a^s(u, \tilde{u}) = \int_{\omega} \mathbf{e}^s(u) \cdot \boldsymbol{\mathfrak{C}}^s \cdot \mathbf{e}^s(\tilde{u}) \, d\mathcal{S}. \tag{8}$$

The shell midsurface can be given as a non-uniform rational B-spline (NURBS) surface for practical applications like outlined in [6] and [18].

3.1.2 Coordinate free Naghdi shell formulation

The traditional Naghdi shell formulation of the previous section uses the covariant coordinates as unknowns. The shell formulation itself can also be written in a coordinate

free formulation which is independent of a coordinate system for the unknowns. With the surface gradient ∇_S defined by

$$\nabla_S \bullet = \Sigma_i \mathbf{a}^i \bullet_{,i}$$

the equivalent of the angle decomposition (1) from plate theory reads $\nabla_S \mathbf{u} \cdot \mathbf{a}_3 = \boldsymbol{\theta} + \boldsymbol{\gamma}$. With the first and second fundamental tensors $\mathcal{A} = a_{ij} \mathbf{a}^i \mathbf{a}^j$ and $\mathcal{B} = b_{ij} \mathbf{a}^i \mathbf{a}^j$ the shell strains may then be expressed by

$$\begin{aligned} 2\varepsilon^m &= \nabla_S \mathbf{u} \cdot \mathcal{A} + \mathcal{A} \cdot (\nabla_S \mathbf{u})^\top, \\ 2\varepsilon^s &= \boldsymbol{\gamma} \mathbf{a}_3 + \mathbf{a}_3 \boldsymbol{\gamma}, \quad 2e^s = \boldsymbol{\gamma}, \\ 2\varepsilon^b &= -\nabla_S \boldsymbol{\theta} \cdot \mathcal{A} - \mathcal{A} \cdot (\nabla_S \boldsymbol{\theta})^\top - \nabla_S \mathbf{u} \cdot \mathcal{B} - \mathcal{B} \cdot (\nabla_S \mathbf{u})^\top. \end{aligned} \quad (9)$$

The elimination of $\boldsymbol{\gamma} = \nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \boldsymbol{\theta}$ changes the shear strain tensor to

$$2\varepsilon^s = (\nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \boldsymbol{\theta}) \mathbf{a}_3 + \mathbf{a}_3 (\nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \boldsymbol{\theta}), \quad 2e^s = \nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \boldsymbol{\theta}. \quad (10)$$

See also [15] for a full derivation (with $\boldsymbol{\theta}$ oriented differently, $\boldsymbol{\theta}_{\text{here}} = -\boldsymbol{\theta}_{\text{there}}$). The bilinear form stays as in (6) and (8).

3.2 The Koiter shell

While the Naghdi shell is the shell equivalent to the Mindlin–Reissner plate, the Koiter shell is the equivalent to the Kirchhoff plate. The additional hypothesis reads

$$\boldsymbol{\theta} = \nabla_S \mathbf{u} \cdot \mathbf{a}_3, \quad \boldsymbol{\gamma} = \nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \boldsymbol{\theta} = 0$$

for shells. It is a valid assumption only for thin shells.

3.2.1 Coordinate free Koiter shell formulation

Direct insertion of the hypothesis into the coordinate free Naghdi shell formulation yields $e^s = 0$ and

$$\begin{aligned} 2\varepsilon^m &= \nabla_S \mathbf{u} \cdot \mathcal{A} + \mathcal{A} \cdot (\nabla_S \mathbf{u})^\top, \\ 2\varepsilon^b &= -\nabla_S (\nabla_S \mathbf{u} \cdot \mathbf{a}_3) \cdot \mathcal{A} - \mathcal{A} \cdot (\nabla_S (\nabla_S \mathbf{u} \cdot \mathbf{a}_3))^\top - \nabla_S \mathbf{u} \cdot \mathcal{B} - \mathcal{B} \cdot (\nabla_S \mathbf{u})^\top \\ &= -\nabla_S \nabla_S \mathbf{u} \cdot \mathbf{a}_3 \cdot \mathcal{A} - \mathcal{A} \cdot (\nabla_S \nabla_S \mathbf{u} \cdot \mathbf{a}_3)^\top \end{aligned}$$

with the chain rule $\nabla_S (\nabla_S \mathbf{u} \cdot \mathbf{a}_3) = \nabla_S \nabla_S \mathbf{u} \cdot \mathbf{a}_3 + \nabla_S \mathbf{u} \cdot \nabla_S \mathbf{a}_3 = \nabla_S \nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \nabla_S \mathbf{u} \cdot \mathcal{B}$. See also [14] and [15, section 5.2] for a full derivation.

3.2.2 Koiter shell in Cartesian coordinates

Writing the Koiter shell in covariant coordinates like the Naghdi shell leads to formulas which are not so easy to implement for general geometries, see [5, section 4.2.3]. A conforming discretisation of this formulation requires C^0 elements for both tangential coordinates and C^1 elements for the normal coordinate.

Alternatively, one can use Cartesian coordinates for the unknowns, see [2]. The coordinate free formulation from above can be reformulated to

$$\begin{aligned} 2\varepsilon^m &= \nabla_S \mathbf{u} \cdot \mathcal{A} + \mathcal{A} \cdot (\nabla_S \mathbf{u})^\top \\ &= \mathbf{u}_{,j} \cdot \mathbf{a}_i + \mathbf{u}_{,i} \cdot \mathbf{a}_j, \\ 2\varepsilon^b &= -\nabla_S \nabla_S \mathbf{u} \cdot \mathbf{a}_3 \cdot \mathcal{A} - \mathcal{A} \cdot (\nabla_S \nabla_S \mathbf{u} \cdot \mathbf{a}_3)^\top \\ &= 2(\mathbf{u}_{,ij} \cdot \mathbf{a}_3 - \Gamma_{ij}^k \mathbf{u}_{,k} \cdot \mathbf{a}_3) \mathbf{a}^i \mathbf{a}^j. \end{aligned}$$

There is no need for further derivative evaluation if the unknowns are given in Cartesian coordinates $\mathbf{u} = u_x \mathbf{e}_x + u_y \mathbf{e}_y + u_z \mathbf{e}_z$. The Cartesian basis vectors $\mathbf{e}_x, \mathbf{e}_y, \mathbf{e}_z$ are fixed in space and thus the derivatives carry over directly to the unknown coordinates themselves;

$$\mathbf{u}_{,i} = \left(\frac{\partial}{\partial \eta^i} u_x\right) \mathbf{e}_x + \left(\frac{\partial}{\partial \eta^i} u_y\right) \mathbf{e}_y + \left(\frac{\partial}{\partial \eta^i} u_z\right) \mathbf{e}_z, \quad \mathbf{u}_{,ij} = \left(\frac{\partial^2}{\partial \eta^i \partial \eta^j} u_x\right) \mathbf{e}_x + \left(\frac{\partial^2}{\partial \eta^i \partial \eta^j} u_y\right) \mathbf{e}_y + \left(\frac{\partial^2}{\partial \eta^i \partial \eta^j} u_z\right) \mathbf{e}_z.$$

This formulation can then easily be implemented. A conforming discretisation requires C^1 finite elements for all three coordinates u_x, u_y, u_z .

3.3 The principle of Endo–Kimura applied to shells

The basic idea of Endo and Kimura can also be applied to the Naghdi shell formulation. The total deformation \mathbf{u} is split into a bending and a shear part with $\mathbf{u} = \mathbf{u}_b + \mathbf{u}_s$. We start from the coordinate free Naghdi formulation of (9) and (10)

$$\begin{aligned} 2\varepsilon^m &= \nabla_S \mathbf{u} \cdot \mathcal{A} + \mathcal{A} \cdot (\nabla_S \mathbf{u})^\top, \\ 2\mathbf{e}^s &= \nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \boldsymbol{\theta}, \\ 2\varepsilon^b &= -\nabla_S \boldsymbol{\theta} \cdot \mathcal{A} - \mathcal{A} \cdot (\nabla_S \boldsymbol{\theta})^\top - \nabla_S \mathbf{u} \cdot \mathcal{B} - \mathcal{B} \cdot (\nabla_S \mathbf{u})^\top \end{aligned}$$

and insert the assumption $\boldsymbol{\theta} = \nabla_S \mathbf{u}_b \cdot \mathbf{a}_3$ to obtain

$$\begin{aligned} 2\varepsilon^m &= \nabla_S \mathbf{u} \cdot \mathcal{A} + \mathcal{A} \cdot (\nabla_S \mathbf{u})^\top, \\ 2\mathbf{e}^s &= \nabla_S \mathbf{u} \cdot \mathbf{a}_3 - \nabla_S \mathbf{u}_b \cdot \mathbf{a}_3, \\ 2\varepsilon^b &= -\nabla_S (\nabla_S \mathbf{u}_b \cdot \mathbf{a}_3) \cdot \mathcal{A} - \mathcal{A} \cdot (\nabla_S (\nabla_S \mathbf{u}_b \cdot \mathbf{a}_3))^\top - \nabla_S \mathbf{u} \cdot \mathcal{B} - \mathcal{B} \cdot (\nabla_S \mathbf{u})^\top \end{aligned}$$

as the shell equivalent to the standard Endo–Kimura plate formulation. Hierarchic and decoupled versions can again be obtained by using the unknowns \mathbf{u} and \mathbf{u}_s with elimination of \mathbf{u}_b for the hierarchic version or by using \mathbf{u}_b and \mathbf{u}_s with elimination of \mathbf{u} for the decoupled version. Just like in the special case of a plate the additional gradient condition leads to a slightly stiffer solution than that of the Naghdi shell.

The decoupled version reads

$$\begin{aligned} 2\boldsymbol{\varepsilon}^m &= \nabla_S(\mathbf{u}_b + \mathbf{u}_s) \cdot \boldsymbol{\mathcal{A}} + \boldsymbol{\mathcal{A}} \cdot (\nabla_S(\mathbf{u}_b + \mathbf{u}_s))^\top, \\ 2\mathbf{e}^s &= \nabla_S \mathbf{u}_s \cdot \mathbf{a}_3, \\ 2\boldsymbol{\varepsilon}^b &= -\nabla_S(\nabla_S \mathbf{u}_b \cdot \mathbf{a}_3) \cdot \boldsymbol{\mathcal{A}} - \boldsymbol{\mathcal{A}} \cdot (\nabla_S(\nabla_S \mathbf{u}_b \cdot \mathbf{a}_3))^\top \\ &\quad - \nabla_S(\mathbf{u}_b + \mathbf{u}_s) \cdot \boldsymbol{\mathcal{B}} - \boldsymbol{\mathcal{B}} \cdot (\nabla_S(\mathbf{u}_b + \mathbf{u}_s))^\top \end{aligned}$$

and the product rule $\nabla_S(\nabla_S \mathbf{u}_b \cdot \mathbf{a}_3) = \nabla_S \nabla_S \mathbf{u}_b + \nabla_S \mathbf{u}_b \boldsymbol{\mathcal{B}}$ together with $\boldsymbol{\mathcal{B}} \cdot \boldsymbol{\mathcal{A}} = \boldsymbol{\mathcal{B}}$ yields the simplified bending strain

$$2\boldsymbol{\varepsilon}^b = -(\nabla_S \nabla_S \mathbf{u}_b \cdot \mathbf{a}_3) \cdot \boldsymbol{\mathcal{A}} - \boldsymbol{\mathcal{A}} \cdot (\nabla_S \nabla_S \mathbf{u}_b \cdot \mathbf{a}_3)^\top - \nabla_S \mathbf{u}_s \cdot \boldsymbol{\mathcal{B}} - \boldsymbol{\mathcal{B}} \cdot (\nabla_S \mathbf{u}_s)^\top.$$

The bending strain features both \mathbf{u}_b and \mathbf{u}_s . Therefore, this formulation is actually not totally decoupled like in the special case of a plate. We still keep the name “decoupled” to link it with the corresponding plate formulation.

We have not yet specified if \mathbf{u}_s is from \mathbb{R}^3 or a smaller subspace. From now on we use $\mathbf{u}_s \in \text{span}\{\mathbf{a}_3\}$ because it is the equivalent of the plate deflection w_s . In contrast, \mathbf{u}_b is from the whole \mathbb{R}^3 because it emulates not only the deflection but also the membrane deformations. It is sensible to use the normal coordinate $\mathbf{u}_s = w_s \mathbf{a}_3$ as an unknown with this assumption. The coordinates of \mathbf{u}_b on the other hand are best expressed in Cartesian coordinates via $\mathbf{u}_b = u_{bx} \mathbf{e}_x + u_{by} \mathbf{e}_y + u_{bz} \mathbf{e}_z$. This yields the covariant coordinates

$$\begin{aligned} 2\varepsilon_{ij}^m &= \mathbf{u}_{b,i} \cdot \mathbf{a}_j + \mathbf{u}_{b,j} \cdot \mathbf{a}_i - 2w_s b_{ij}, \\ 2e_i^s &= w_{s,i}, \\ 2\varepsilon_{ij}^b &= -2\mathbf{u}_{b,ij} \cdot \mathbf{a}_3 + 2\sum_k \Gamma_{ij}^k \mathbf{u}_{b,k} \cdot \mathbf{a}_3 + 2w_s c_{ij} \end{aligned}$$

which can be used for implementing the decoupled Endo–Kimura shell with C^1 elements for the three Cartesian coordinates of \mathbf{u}_b and C^0 elements for the normal coordinate w_s of \mathbf{u}_s .

3.4 Hierarchic Naghdi shell formulation based on Endo–Kimura

A relaxation of the decoupled Endo–Kimura shell formulation to obtain again the Naghdi shell solution can be done analogously to the relaxation of the according plate model

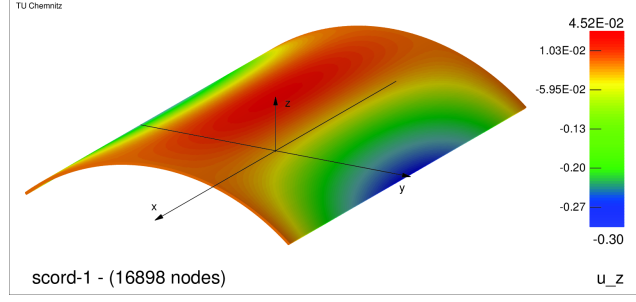


Figure 5: Scordelis-Lo roof with vertical deflection u_z^{3D}

in 2.10. Adding the relaxation angle $\boldsymbol{\rho} = \rho_1 \mathbf{a}^1 + \rho_2 \mathbf{a}^2$ with $\boldsymbol{\theta} = \nabla_S \mathbf{u}_b \cdot \mathbf{a}_3 - \boldsymbol{\rho}$ and $\boldsymbol{\gamma} = \nabla_S \mathbf{u}_s \cdot \mathbf{a}_3 + \boldsymbol{\rho}$ results in the shear and bending strains

$$\begin{aligned} 2e^s &= \nabla_S \mathbf{u}_s \cdot \mathbf{a}_3 + \boldsymbol{\rho}, \\ 2\varepsilon^b &= -(\nabla_S \nabla_S \mathbf{u}_b \cdot \mathbf{a}_3) \cdot \mathcal{A} - \mathcal{A} \cdot (\nabla_S \nabla_S \mathbf{u}_b \cdot \mathbf{a}_3)^\top - \nabla_S \mathbf{u}_s \cdot \mathcal{B} - \mathcal{B} \cdot (\nabla_S \mathbf{u}_s)^\top \\ &\quad + \nabla_S \boldsymbol{\rho} \cdot \mathcal{A} + \mathcal{A} \cdot (\nabla_S \boldsymbol{\rho})^\top \end{aligned}$$

with the covariant coordinates

$$\begin{aligned} 2e_i^s &= w_{s,i} + \rho_i, \\ 2\varepsilon_{ij}^b &= -2\mathbf{u}_{b,ij} \cdot \mathbf{a}_3 + 2\Sigma_k \Gamma_{ij}^k \mathbf{u}_{b,k} \cdot \mathbf{a}_3 + 2w_s c_{ij} + \rho_{i,j} + \rho_{j,i} - 2\Sigma_k \Gamma_{ij}^k \rho_k. \end{aligned}$$

The membrane strains are the same as above. A conforming discretisation may be achieved with C^1 elements for the three Cartesian coordinates of \mathbf{u}_b and C^0 elements for the normal coordinate w_s of \mathbf{u}_s and the two covariant coordinates of $\boldsymbol{\rho}$.

3.5 Numerical example: Scordelis–Lo roof

The Scordelis–Lo roof was used to validate the exactness of the models and the implementation. The problem with the vertical deflection u_z^{3D} for is illustrated in Figure 5. A roof section cut out from a right circular cylinder with an opening angle of 80° rests on a rigid diaphragm with its curved edges. The normal and tangential deformation with respect to the spanning circle are zero at the curved edges (together with their first derivatives in tangential direction); the straight edges are free. The shell thickness is 0.25, the cylinder radius 25 and the length 50. Isotropic material with $E = 4.32 \cdot 10^8$ and $\nu = 0$ is used and a vertical load per unit surface of -90 is applied. The absolute value of the maximum vertical deflection of $\max |u_z^{3D}| = 0.3024$, which occurs at the middle of the free edges, is suggested to be used as a benchmark in [10]. Our own overkill solution with 137962 adaptively refined biquadratic Lagrangian elements obtained $\max |u_z^{3D}| = 0.3043$. We did not make use of symmetries and simulated the whole domain.

#El	Naghdi bilin.		Naghdi biquadr.		EK shell		hier. Naghdi/EK	
	max $ u_z^{3D} $	#it	max $ u_z^{3D} $	#it	max $ u_z^{3D} $	#it	max $ u_z^{3D} $	#it
4	2.02E-3	9	3.74E-2	36	0.2138	77	0.2809	29
16	7.72E-3	34	0.1472	267	0.2959	643	0.2960	653
64	1.99E-2	108	0.2829	661	0.3025	1011	0.3025	1047
256	3.76E-2	255	0.3007	545	0.3030	1065	0.3031	744
1024	7.27E-2	587	0.3026	423	0.3030	605	0.3032	978
4096	0.1485	1184	0.3038	452	0.3030	709	0.3034	1629
16384	0.2362	1764	0.3043	534	0.3030	670	0.3036	2788

Table 10: Results and iteration numbers for Scordelis–Lo roof

#El	Naghdi biquadr.		EK shell		hier. Naghdi/EK	
	max $ u_z^{3D} $	#it	max $ u_z^{3D} $	#it	max $ u_z^{3D} $	#it
4	3.77E-3	36	0.1707	59	0.2983	30
16	2.16E-2	271	0.2181	1317	0.2184	1347
64	0.1107	1310	0.3067	5382	0.3067	6071
256	0.2404	2949	0.8183	9494	0.3183	9655
1024	0.3020	2712	0.3204	5615	0.3205	5636
4096	0.3022	501	0.3206	3019	0.3206	3050
16384	0.3197	3426	0.3206	4545	0.3206	5272

Table 11: Results and iteration numbers for thinner Scordelis–Lo roof

Results obtained with uniform refinement of bilinear as well as biquadratic Lagrangian elements for the standard Naghdi formulation and a combination of bilinear Lagrangian and Bogner–Fox–Schmit elements for the C^0 and C^1 parts of the Endo–Kimura shell formulation as well as the hierarchic Naghdi shell formulation based on Endo–Kimura are collected in Table 10. Iteration numbers of all methods are much higher than for their counterparts in the plate example. The iteration numbers of the Koiter shell in cartesian coordinates with BFS elements which are not shown in the table are about the same as those of the Endo–Kimura shell. The bilinear elements show slow h -convergence but the h -convergence of the other methods appears satisfying.

In order to conduct a locking study we also considered a thinner Scordelis–Lo roof with thickness 0.025 and load -0.9 . Results are collected in Table 11. We observe that both Endo–Kimura shell formulations have the best h -convergence but at extreme iteration numbers. A better preconditioner tailored for these formulations which might reduce these iteration numbers would be desirable. The Koiter shell in cartesian unknowns with BFS elements shows iteration numbers in the same order of magnitude.

4 Conclusion

We have compared several known and two new plate formulations with respect to their locking behaviour and their numerical efficiency combined with a conforming discretisation. The two new formulations for the Endo–Kimura variant and the original Mindlin–Reissner plate turned out to be locking free and among the most efficient methods for both thick and thin plates.

Extensions of both formulations to shells have been presented. The example problem has shown that the solution of shell problems is more challenging and a better preconditioner might be needed. Nevertheless both methods are locking free and thus suitable for thick and thin shells.

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